# An Attempt of Simulating the Real Time Filling of H<sub>2</sub> Cylinder at 70MPa

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This report presents the computational fluid dynamics (CFD) studies on real time filling of  $H_2$  gas in cylindrical tank at 70MPa. A standard type IV, 39.1L vessel composed of plastic and Carbon Fibered Reinforced Plastic (CFRP) for liner and laminate is used in the present study. A density based algorithm of ANSYS Fluent 13.0 is applied to solve the time dependent 3D RANS equations with SST *k-w* turbulence model. User defined real gas model is used for density computation to model compressibility effects. The energy equation for the working gas is coupled to the energy equation for the solid walls in order to estimate the convective heat transfer from the compressed gas to tank wall. Duration of the tank filling of 163 s which is almost the same to real time filling at refueling station, is used in the present work.

Key Words : compressible flow, convection, hydrogen tank, numerical simulation, real gas

# 1. Introduction

As a sustainable and green energy carrier, hydrogen gas can meet the current needs of energy without sacrificing the cultural and economic growth, the environment (global worming, climate change, environmental degradation, etc.) or the ability of future generations to live even better than we do today [1]. Hydrogen powered vehicles highlight the need of storing hydrogen ( $H_2$ ) gas for its clean and efficient conversion to energy. At present, pressurised gas cylinders represent the simplest and most mature technology for  $H_2$  storage whose feasibility has already been demonstrated at working pressure up to 70MPa [2]. Pressurised cylinders are also the preferred near-term option for direct-hydrogen and hybrid fuel cell vehicles. Refuelling vehicle cylinders involves transferring high-pressure hydrogen gas from the fuelling station tanks through a dispenser into the vehicle cylinder. The dispenser controls the rate of filling, ensures safe operation, and delivers the rated mass of gas [2]. The total time required to refuel a cylinder is important for consumer acceptance and it can have significant financial consequences in commercial applications. However, rapid filling of hydrogen increases the gas temperature. Current regulation imposes limits on the maximum average gas temperature allowed within the cylinders for vehicular applications; the average  $H_2$  gas temperature cannot exceed 85°C [4-6]. Therefore, monitoring and feedback of temperature data inside the vessel and on the tank wall are essential during the real time filling.

Some experimental and numerical works may be found in literature. To evaluate temperature distribution within the cylinder wall, a simple approach consists in coupling a single gas temperature evolution prediction and a one-dimensional conduction calculation within the wall [7]. Such a method was also applied in Air Liquide R&D [8] and allowed to understand the influence of filling parameters. Experimental studies using measurement devices placed inside the vessel have also been carried out, allowing evaluating the temperature in few points [8-10] or in a more extended region using a large matrix of sensors [3]. CFD simulations have also been performed to 35MPa filling for a duration inferior to one minute [11-13]. In the present work, CFD simulations are conducted on the real time filling of  $H_2$  gas in cylindrical tank at 70MPa. Real gas effects are incorporated by using user defined real gas model to account for the deviation of hydrogen gas from ideal gas behaviour. Filling time of 163 s that mostly needed at practical refuelling station is used as the duration of filling in the present work.

2. Numerical model and computational methodology

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# 2.1 Governing equations

The simulation of real time filling of hydrogen gas in pressurized cylindrical vessel requires the solution of a system of governing equations in the unsteady form. The governing equations used in the present work are time dependent compressible Reynolds and Favre averaged Navier-Stokes (RANS) equations. For the closure of the non-linear term from the convective acceleration and Reynolds stress, the well known SST k- $\omega$  turbulence model [14-16] is used.

The conservation of mass and momentum do not apply at the tank walls. The energy equation for the walls is given by;

$$\frac{\partial}{\partial t}(\rho h) = \nabla \cdot \left(k_s \nabla T\right) \tag{1}$$

where the sensible enthalpy is given by;

$$h = \int_{T-r}^{T} c_{ps} dT \tag{2}$$

The energy equation for the working gas is coupled to the energy equation for the solid walls.

In real time filling of hydrogen tank involving high pressure and temperature compressibility effects are significant, and the behavior of hydrogen gas is expected to deviate from ideal gas. Ideal gas simulation may not be sufficient, and therefore, Sakoda et al. [17] equation of state (EOS) is used in the present work for real gas modeling. The functional form of the virial EOS proposed by Sakoda et al. [16] can be expressed by,

$$Z = 1 + B(T)\rho + C(T)\rho^{2} + D(T)\rho^{3}$$
(3)

$$B(T) = b_1 + b_2 \exp(T_r^{-1}) + b_3 T_r^{-k_1} + b_4 T_r^{-k_2}$$
(4)

$$C(T) = c_1 + c_2 T_r^{-k_3} + c_3 T_r^{-k_4}$$
(5)

$$D(T) = d_1 T_r^{-k_5} \tag{6}$$

In the above equations, *B*, *C* and *D* are the second, third, and forth virial coefficients, respectively, and  $T_r = T/T_c$ , where  $T_c$  is the critical temperature (33.145K for H<sub>2</sub>). The coefficients  $k_1 - k_5$ ,  $b_1 - b_4$ ,  $c_1 - c_3$ , and  $d_1$  were determined using non-linear least-squares fitting to minimize the sum of squares,

$$SS = \sum_{i=1}^{I} w_i^2 \left( Z_{Exp,i} - Z_{Cal,i} \right)^2 + \sum_{j=1}^{J} w_j^2 \left( B_{Exp,j} - B_{Cal,j} \right)^2 + \sum_{k=1}^{K} w_k^2 \left( C_{Exp,k} - C_{Cal,k} \right)^2$$
(7)

where *w* is the weight, *I*, *J*, and *K* are the number of density data, the second virial coefficients, and the third virial coefficients, respectively. The determined coefficients are listed in Table 1. The specific heat,  $c_{pg}$ , thermal conductivity,  $k_{g}$ , speed of sound, *a*, surface tension,  $\sigma$  and viscosity,  $\mu$  are computed employing equations proposed by Leachmann et al. [18].

|       |                             |       | 1                          |
|-------|-----------------------------|-------|----------------------------|
| $b_1$ | 0.161 777×10 <sup>-1</sup>  | $d_1$ | $0.716\ 809{	imes}10^{-4}$ |
| $b_2$ | -0.102 798×10 <sup>0</sup>  | $k_1$ | 0.0612                     |
| $b_3$ | $0.129\ 501{	imes}10^{0}$   | $k_2$ | 2.71                       |
| $b_4$ | 0.628 764×10 <sup>-1</sup>  | $k_3$ | 2.53                       |
| $c_1$ | 0.288 234×10 <sup>-3</sup>  | $k_4$ | 4.85                       |
| $c_2$ | 0.203 867×10 <sup>-2</sup>  | $k_5$ | 1.09                       |
| $c_3$ | -0.779 969×10 <sup>-3</sup> |       |                            |

Table 1 Coefficients in the virial EOS for Eqs. 3-6

#### 2.2 Numerical methods

The density based solver of ANSYS Fluent 13.0 [19] is used to solve coupled three dimensional system of equations. Hydrogen gas is selected as working fluid from ANSYS Fluent 13.0 database. Spatial derivatives are discretized by 3rd order MUSCL scheme that is a blend of central differencing and second-order upwind schemes. Second-order central difference scheme is used for viscous terms. Second-order implicit time integration scheme is used to discretize the temporal derivatives. Convective heat transfer from the gas to tank wall is estimated by wall discritization for local heat transfer coefficient. Radiation heat transfer between the gas and wall is assumed to be negligible since temperature difference between the gas and wall is very small.

#### 2.2 Computational conditions

Schematics of the computational geometry and grids are presented in Fig. 1. Type IV vessel having the volume of 39.1L, inner radius of 0.28mm, liner and laminate thicknesses are of 3 and 22 mm, respectively, is used in the present computational work. Model geometry is divided into two domains: fluid domain filled with hydrogen gas and solid domain involving liner and laminate regions on the tank wall and inlet tube. Liner is made of plastic and laminate is constructed from carbon fiber reinforced plastics (CFRP). Properties of liner and laminate materials used in the computation are shown in Table 2. Inlet tube is protruding into the tank up to a distance of 70 mm. Structured clustered grid is used consisted of 222,752 hexahedral elements in fluid domain and 109,156 hexahedral elements in solid domain. Grids are densely clustered in near wall regions to capture flow features in boundary layers. Enough nodes are located in the structured grid in the solid region to compute wall heat transfer accurately.

Pressure inlet boundary condition is specified at tube inlet. Inlet pressure and temperature are varied with time, as shown in Fig.2, to replicate the experimental conditions. Curve fitted experimental data of pressures and temperatures are read into the numerical model as a UDF (user defined function). No slip boundary conditions are specified at inner walls. Energy equation for gas is coupled to energy equation for walls. A constant heat transfer coefficient of 10 W/m<sup>2</sup> K is specified at the outer wall. As the filling completed in a couple of minutes, a constant ambient temperature of 283.65 K is specified. Pressure and temperature of the gas is assumed to be uniform within the tank before filling. The flow field is initialized with initial conditions of hydrogen gas in the vessel i.e. pressure 1.32MPa and temperature 279.83 K. Initial temperature of the tank walls is assumed to be same as the initial temperature of gas.

For optimum time step size, an initial guess  $\Delta T=10^4$ s, is made based on the initial gas velocity and minimum cell size.

|              |          | Material | Specific heat (J/kg.K) | Thermal conductivity (W/m.K) | Density (kg/m <sup>3</sup> ) |
|--------------|----------|----------|------------------------|------------------------------|------------------------------|
| Type IV tank | Liner    | plastic  | 1578                   | 1.17                         | 1286                         |
|              | Laminate | CRFP     | 1075                   | 1.14                         | 1375                         |
|              |          |          |                        |                              |                              |

Table 2 Properties of liner and laminate materials



Fig. 1 Schematics of tank geometry and computational grids



Fig. 2 Inlet pressure and temperature

Subsequently, the solution is reiterated using reduced time step sizes. For grid convergence investigation, initially hexahedral coarse mesh is used and then a finer mesh with double the original mesh density is created. The grid is also refined in the shear layer between the incoming jet and gas already present in the tank. The problem is reiterated using the time step size of  $\Delta T=10^{-4}$ s and refined mesh. However, shorter time steps and finer meshes turned out to be computationally expensive. Therefore, the selected time step and mesh size are used in all subsequent simulations.

# 3. Results and discussion

Velocity vectors along the tank mid plane (*xy*-plane) at 8 s and 12 s from the commencement of filling are presented in Figs. 3(a) and 3(b), respectively. Gas velocity and velocity gradients are found to be highest in the vicinity of tank axis. It is observed that the incoming jet strikes the bottom of the tank, and flow turns along the rear surface towards the inlet forming a secondary flow region on either sides of inlet jet. These secondary flow regions shift positions and vary in size with respect to time during the fill. Flow velocity gradually decreases, as shown in Fig. 3(b), with the increase in pressure as density of gas increases as the tank gets filled up. Flow at the inlet is turbulent throughout. Development of flow field within the tank is dominated by the structure of turbulent jet flowing from the inlet. More turbulence near the inlet results in enhanced heat transfer to the wall in this region.

Contours showing in Fig. 4 are illustrated the variations of density along the mid plane of the flowfield. From the contour sequences, it is observed that density of gas increases with the increase in tank pressure. A nearly conical inlet jet plume at transonic speed can be observed and initially, it extends from the inlet to about tank mid-section in the longitudinal direction; while the jet plume spreads about the mid-section in the radial direction. The plume increases initially and decreases again in size as filling proceeds due to compression of gas and 'growth' of the bulk region. However, the jet plume shows symmetric characteristics during filling of tank.

Temperature variation during filling is presented in Fig. 5. As per the results, tank may be divided into two regions: the near inlet region with a high temperature gradient and a bulk flow region with no significant change in temperature. A nearly conical inlet gas plume can be observed and initially, it extends from the inlet to about tank mid section in the longitudinal direction. Local temperatures in the plume region are significantly lower than mean gas temperature. High temperature gradient in the vicinity of inlet is due to the influence of the cooler incoming gas. Also high level of turbulence due to







Fig. 4 Density contours on *xy*-plane



Fig. 5 Temperature contours on xy-plane



Fig. 6 Temperature distribution on vessel wall

recirculating gas mixing with high velocity inlet jet facilitates a high rate of heat transfer to the walls. Temperature is more or less same outside the inlet plume region.

Figures 6(a) and 6(b) show the temperature-time histories along wall at the front end and rear end of tank, respectively. High temperature gradient is experienced in the liner, while the temperature gradient gradually diminishes along the width, and reaches to zero after one-third of the laminate section. Due to the low thermal conductivity of CFRP, temperature of outer surface of the laminate has not recorded any appreciable change at any location during filling. However, comparing with the other locations a significant increase in temperature is recorded at the rear end of the tank.

# 7. Conclusion

A 3D numerical model for simulating the real time filling of hydrogen gas into cylindrical tank at 70MPa has been developed. Thermo-fluid dynamic characteristics of hydrogen gas during filling have been investigated. Time dependent compressible RANS equations were solved along with SST k- $\omega$  turbulence model. Real gas equation of state was considered to accurately compute the density and temperature of the gas flowfield. The numerical model predicted the temperature field with a good accuracy. A conical jet plume was observed near tube exit region initially, and found extending and spreading to about tank mid-section in the longitudinal and radial direction, respectively. Size of the jet plume was decreasing in size as filling proceeds, and found increasing of bulk region. Temperature distribution along tank wall was obtained.

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